

# TURBULENT EXCHANGE OF MOMENTUM, MASS, AND HEAT BETWEEN FLUID STREAMS AND PIPE WALL

DARSHANLAL T. WASAN and CHARLES R. WILKE

Lawrence Radiation Laboratory and Department of Chemical Engineering,  
University of California, Berkeley, California

(Received 28 May 1963)

**Abstract**—A new correlation is presented to describe mass and heat transfer to a fluid in a fully developed turbulent flow in a pipe. The correlation differs from earlier empirical relations in that it is based on a theoretical continuous eddy-viscosity distribution from the wall to the center of the pipe. Transfer rates calculated from the new correlation are in excellent agreement with experimental data on mass and heat transfer to fluid streams.

## NOMENCLATURE

$C$ ,	concentration of diffusing species;
$C_w$ ,	concentration at the wall;
$C_{avg}$ ,	time-average concentration;
$C_p$ ,	heat capacity;
$D$ ,	molecular diffusivity;
$F$ ,	functional notation;
$g_e$ ,	conversion constant;
$k_c$ ,	mass-transfer coefficient;
$Nu$ ,	Nusselt number;
$Pr$ ,	Prandtl number;
$Sc$ ,	Schmidt number;
$St$ ,	Stanton number;
$T_w$ ,	temperature at the wall;
$T$ ,	temperature;
$T_{avg}$ ,	average temperature;
$U^+$ ,	dimensionless velocity;
$y$ ,	distance from the wall;
$y^+$ ,	$yU_\tau/\nu$ ;
$\mu$ ,	molecular viscosity;
$\nu$ ,	kinematic viscosity;
$\rho$ ,	density of fluid;
$\epsilon_v$ ,	eddy viscosity;
$\epsilon_d$ ,	eddy diffusivity for mass;
$\epsilon_c$ ,	eddy conductivity;
$\tau_w$ ,	shear stress at the wall;
$St_m$ ,	Stanton number for mass;
$St_h$ ,	Stanton number for heat;
$u$ ,	fluctuating velocity in the axial direction;
$v$ ,	fluctuating velocity in the radial direction;

$\bar{uv}$ ,

Reynolds stress.

## INTRODUCTION

THE turbulent exchange of momentum, mass, and heat in the vicinity of a boundary is encountered in many engineering processes. It is well established [1, 2, 3] that in the vicinity of a boundary the turbulent exchange of momentum, mass, and heat is governed not only by molecular motion but also by an eddy motion. To predict rates of mass and heat transfer between a moving fluid and a wall it is therefore essential to understand the mechanism of eddy motion in the vicinity of a wall. Unfortunately very little is known about the distribution of eddies very close to the wall. In the past, due to the absence of both theoretical analyses and experimental data for the region close to the wall, a number of empirical expressions have been proposed for eddy-viscosity distributions near a wall. An excellent review of several such existing correlations for fully developed turbulent flow in a pipe with constant fluid properties has been presented by Sherwood [3].

Recently Wasan, Tien, and Wilke [4] pointed out that most of the proposed eddy-viscosity distributions do not satisfy the theoretical criterion which states that the turbulent contribution to Reynolds stress  $\bar{uv}$  near the wall is proportional to  $y^n$  where  $n$  is not less than three. This criterion was first derived by Townsend [5]. Also, all the previous analyses are based on the

concept of three sharply defined fluid layers, namely laminar sublayer, buffer, and turbulent layers. However, according to these authors [4], and also Gowariker [6], this concept of three different fluid layers leads to an unrealistic discontinuity in the value of the eddy-viscosity function with respect to that obtained from logarithmic distribution in the turbulent core. Rannie [7] and Sleicher [8] in their analyses avoided this point of discontinuity, but the eddy-viscosity functions of both of these authors do not give satisfactory relationships for analogy expressions for heat- and mass-transfer rates for systems with high Schmidt or Prandtl numbers. From velocity-variation data and from turbulent shear-stress data of Laufer [9] it is evident that the degree of turbulence in the moving fluid varies continuously from the wall to the axis of a pipe. Hence, the concept of three distinct fluid layers would appear to be incorrect.

By using the equations of mean motion and the well established empirical logarithmic velocity distribution in the turbulent core, Wasan, Tien, and Wilke [4] have derived theoretical expressions for the continuous variation of velocity and eddy viscosity for the wall region of pipe flow. The distributions of these authors fit the experimental data on velocity and turbulent shear stress over the wall region. Their velocity and eddy-viscosity distributions for the wall region ( $0 \leq y^+ \leq 20$ ) are presented in equations (1) and (2) as follows:

$$U^+ = y^+ - 1.04 \times 10^{-4} y^{+4} + 3.03 \times 10^{-6} y^{+5}, \quad (1)$$

and

$$\frac{\epsilon_v}{\nu} = \frac{4.16 \times 10^{-4} y^{+3} - 15.15 \times 10^{-6} y^{+4}}{1 - 4.16 \times 10^{-4} y^{+3} + 15.15 \times 10^{-6} y^{+4}} \quad (2)$$

We present correlations relating the fluid friction and turbulent-exchange rates of mass and heat over a wide range of Schmidt and Prandtl numbers based on these distributions. The corresponding concentration and temperature distributions for the wall region of pipe flow are also presented.

#### ANALYSIS

Consider a fully developed turbulent flow of fluid with constant properties in a pipe having

walls kept at a constant concentration  $C_W$  or at a constant temperature  $T_W$ . Mass or heat is transferred to the fluid stream both by molecular and by eddy motions. As is customary, the shear-stress, mass-transfer, and heat-transfer fluxes can be written as the sum of molecular and turbulent fluxes as follows:

$$\tau g_c = (\mu + \rho \epsilon_v) \frac{dU}{dy}, \quad (3)$$

$$N_A = (D + \epsilon_d) \frac{dC}{dy}, \quad (4)$$

and

$$q = \rho c p(k + \epsilon_c) \frac{dT}{dy}, \quad (5)$$

where  $\tau$  is the shear stress at a plane parallel to the wall,  $N_A$  is the mass flux, and  $q$  is the heat flux across a plane parallel to the wall.

To solve (3), (4) and (5), several assumptions are usually made [3]. First, in the wall region the shear stress  $\tau$ , mass-transfer flux  $N_A$ , and heat-transfer flux  $q$  are assumed to be constant and equal to the value at the wall. Second, in the fully turbulent region the variation of shear stress, mass transfer, and heat transfer is such that  $\tau/N_A$  and  $\tau/q$  are constant, and molecular viscosity  $\mu$ , molecular diffusivity  $D$  and thermal conductivity  $k$  can be neglected. Third, in the case of mass transfer the interfacial velocity is assumed to be negligible. The last important assumption is that eddy diffusivities of momentum  $\epsilon_v$ , mass  $\epsilon_d$  and heat  $\epsilon_c$  are equal.

Now, by combinations of (3) and (4) and after integration, for the turbulent region one obtains

$$\frac{C_{avg} - C_1}{U_{avg} - U_1} = \frac{\rho N_{AW}}{\tau_W g_c}, \quad (6)$$

where  $C_{avg}$  and  $U_{avg}$  refer to the average concentration and velocity respectively, and  $C_1$  and  $U_1$  refer to concentration and velocity each corresponding to  $y^+ = 20$ . After combining (1) and (6) one gets

$$C_{avg} - C_1 = \frac{\rho}{\tau_W g_c} N_{AW} U_{avg} [1 - 13.0 \sqrt{(f/2)}], \quad (7)$$

where the concentration  $C_1$  corresponding to  $y^+ = 20$  is determined as follows:

In the wall region equation (4) can be re-written as

$$N_{AW} = \sqrt{\left(\frac{\tau_w g_c}{\rho}\right)} \left(\frac{1}{Sc} + \frac{\epsilon_v}{v}\right) \frac{dC}{dy^+}. \quad (8)$$

Integration of (8) gives

$$C_1 - C_W = \frac{N_{AW}}{\sqrt{(\tau_w g_c / \rho)}} \int_0^{20} \frac{dy^+}{(1/Sc + \epsilon/v)}. \quad (9)$$

Therefore, from (7) and (9) there results

$$\begin{aligned} C_{avg} - C_W &= \frac{\rho N_{AW}}{\tau_w g_c} U_{avg} [1 - 13.0 \sqrt{(f/2)}] \\ &+ \frac{N_{AW}}{\sqrt{(\tau_w g_c / \rho)}} \int_{y^+}^{20} \frac{dy^+}{(1/Sc + \epsilon/v)}. \end{aligned} \quad (10)$$

But the mass flux  $N_{AW}$  at the wall can be given by the expression

$$N_{AW} = k_c (C_{avg} - C_W). \quad (11)$$

Hence, from (10) and (11) the expression for mass-transfer Stanton number in terms of Schmidt number and friction factor becomes

$$\begin{aligned} St_m &= \frac{k_c}{U_{avg}} \\ &= \frac{f/2}{1 + \sqrt{(f/2)} \left[ \int_0^{20} \frac{dy^+}{(1/Sc + \epsilon/v)} - 13.0 \right]}, \end{aligned} \quad (12)$$

where  $\epsilon/v$  is given by equation (2).

Similarly from (3) and (5) the expression for heat-transfer Stanton number in terms of Prandtl number and friction factor can be obtained as

$$St_h = \frac{f/2}{1 + \sqrt{(f/2)} [F(Pr, 20) - 13.0]}. \quad (13)$$

#### CONCENTRATION AND TEMPERATURE DISTRIBUTIONS

When (8) and (10) are combined, the expression for the concentration distribution becomes

$$\frac{C - C_W}{C_{avg} - C_W} = \frac{F(Sc, y^+)}{F(Sc, 20) + \sqrt{(2/f)} - 13.0}. \quad (14)$$

Similarly the temperature distribution is obtained as

$$\frac{T - T_W}{T_{avg} - T_W} = \frac{F(Pr, y^+)}{F(Pr, 20) + \sqrt{(2/f)} - 13.0}, \quad (15)$$

where

$$F(Sc, y^+) = \int_0^{y^+} \frac{dy^+}{[1/Sc + \epsilon/v(y^+)]}. \quad (16)$$

Equations (14), (15) and (16) give the concentration and temperature distributions in the wall region of a pipe flow.

The functions  $F(Sc$  or  $Pr, y^+$ ) and  $F(Sc$  or  $Pr, 20$ ) appearing in (13) and (15) involve the integrals which cannot be solved in closed forms. These were obtained by numerical integration using Simpson's one-third rule with a digital computer. These functions are shown in Table 1 for  $Sc$  or  $Pr$  numbers over a range of 0.1 to 10 000.

#### DISCUSSION

In Fig. 1 the function  $F(Sc, y^+)$  or  $F(Pr, y^+)$  is shown for the range of Schmidt and Prandtl

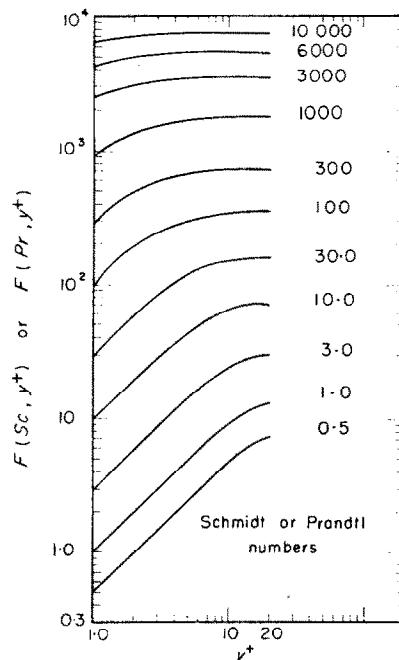
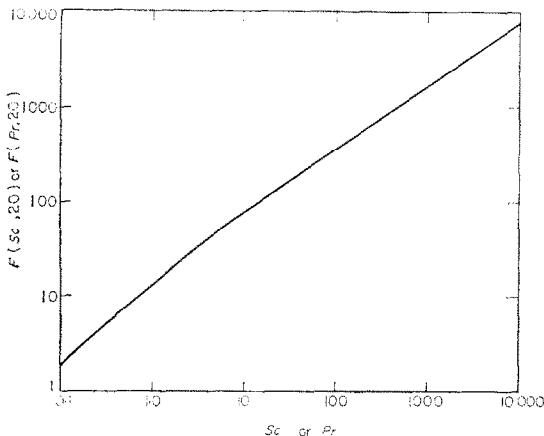


FIG. 1. Plot of  $F(Sc, y^+)$  or  $F(Pr, y^+)$  vs.  $y^+$ .

Table 1. Tabulation of functions  $F(Sc, y^+)$  or  $F(Pr, y^+) = \int_0^{y^+} \frac{dy^+}{\frac{1}{Sc} + \frac{\epsilon}{\nu}(y^+)} =$

$Sc$ or $Pr$	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20
0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1.0	1.1	1.2	1.3	1.4	1.5	1.6	1.7	1.8	1.9	1.8	
0.2	0.4	0.6	0.8	1.0	1.2	1.4	1.6	1.8	2.0	2.1	2.3	2.5	2.7	2.8	3.0	3.1	3.2	3.3	3.3	
0.3	0.6	0.9	1.2	1.5	1.8	2.1	2.4	2.8	3.1	3.5	3.9	4.2	4.5	4.8	5.1	5.4	4.6	4.7	4.8	
0.4	0.8	1.2	1.6	2.0	2.4	2.8	3.1	3.5	3.9	4.2	4.8	5.2	5.6	6.0	6.4	6.8	7.2	7.3	7.4	
0.5	1.0	1.5	2.0	2.5	3.0	3.4	3.9	4.4	4.8	5.2	5.6	6.0	6.3	6.7	7.0	7.4	7.8	8.5	8.6	
0.6	1.2	1.8	2.4	3.0	3.6	4.1	4.7	5.2	5.7	6.2	6.7	7.2	7.4	7.8	8.2	8.4	8.8	9.6	9.7	
0.7	1.4	2.1	2.8	3.5	4.1	4.8	5.4	6.0	6.6	7.2	7.7	8.1	8.5	8.8	9.1	9.3	9.5	10.8	10.8	
0.8	1.6	2.4	3.2	4.0	4.7	5.5	6.2	6.9	7.5	8.1	8.6	9.1	9.6	10.1	10.6	11.0	11.3	11.7	11.9	
0.9	1.8	2.7	3.6	4.5	5.3	6.1	6.9	7.7	8.4	9.0	9.6	10.1	10.6	11.0	11.5	11.7	11.8	12.8	13.0	
1.0	2.0	3.0	4.0	4.9	5.9	6.8	7.7	8.5	9.3	10.0	10.6	11.1	11.6	12.0	12.6	13.0	13.6	13.8	14.0	
1.1	2.2	3.3	4.4	5.4	6.5	7.5	8.4	9.3	10.1	10.9	11.5	12.1	12.6	13.0	13.3	13.6	13.8	13.9	14.0	
1.2	2.4	3.6	4.8	5.9	7.0	8.1	9.1	10.1	11.0	11.8	12.5	13.1	13.6	14.0	14.5	14.8	14.9	15.0	15.0	
1.3	2.6	3.9	5.2	6.4	7.6	8.8	9.9	10.9	11.8	12.6	13.4	14.0	14.5	15.0	15.5	15.8	15.9	16.0	16.0	
1.4	2.8	4.2	5.6	6.9	8.2	9.4	10.6	11.6	12.6	13.5	14.3	14.9	15.5	16.3	16.5	16.7	16.9	17.0	17.0	
1.5	3.0	4.5	5.9	7.4	8.8	10.1	11.3	12.4	13.4	14.2	15.0	15.8	16.4	17.0	17.5	17.8	17.9	17.9	17.9	
1.6	3.2	4.8	6.3	7.9	9.3	10.7	12.0	13.2	14.3	15.2	16.0	16.7	17.3	17.8	18.1	18.4	18.7	18.8	18.8	
1.7	3.4	5.1	6.7	8.3	9.9	11.3	12.7	13.9	15.1	16.0	16.9	17.6	18.2	18.7	19.0	19.3	19.5	19.7	19.8	
1.8	3.6	5.4	7.1	8.8	10.4	12.0	13.4	14.7	15.9	16.9	17.5	18.5	19.1	19.6	20.2	20.4	20.6	20.7	20.7	
1.9	3.8	5.7	7.5	9.3	11.0	12.6	14.1	15.4	16.6	17.7	18.6	19.3	19.9	20.4	20.8	21.1	21.4	21.5	21.5	
2.0	4.0	6.0	7.9	9.8	11.6	13.4	14.8	17.4	18.5	19.4	20.2	21.0	21.6	22.1	22.5	22.8	23.0	23.2	23.4	
2.1	4.2	6.3	8.3	10.3	12.1	13.9	15.5	16.9	18.2	19.3	20.1	21.0	21.8	22.5	23.0	23.4	23.7	24.0	24.1	
2.2	4.4	6.6	8.7	10.7	12.7	14.5	16.2	17.6	18.4	19.7	20.9	21.8	22.7	23.3	23.8	24.2	24.7	24.8	24.9	
2.3	4.6	6.9	9.1	11.2	13.2	15.1	16.8	18.4	19.7	20.9	21.5	22.6	23.5	24.1	24.6	25.1	25.5	25.8	25.8	
2.4	4.8	7.2	9.5	11.7	13.8	15.7	17.5	19.1	20.5	21.5	22.4	23.4	24.3	24.9	25.4	25.8	26.3	26.5	26.6	
2.5	5.0	7.5	9.9	12.2	14.3	16.4	18.2	19.8	21.2	22.4	23.4	24.3	25.3	26.2	26.6	26.9	27.1	27.3	27.4	
2.6	5.2	7.7	10.2	12.6	14.9	17.0	18.8	20.5	21.9	23.2	24.2	25.0	25.7	26.5	27.0	27.4	27.7	28.1	28.2	
2.7	5.4	8.0	10.6	13.1	15.4	17.6	19.5	20.5	21.2	22.7	23.9	25.0	25.7	26.6	27.3	27.8	28.1	28.9	29.0	
2.8	5.6	8.3	11.0	13.6	16.0	18.2	19.7	21.2	21.9	23.4	24.7	25.5	26.5	27.4	28.1	28.7	29.3	29.8	29.8	
2.9	5.8	8.6	11.4	14.0	16.5	18.8	20.8	22.6	23.4	24.1	25.4	26.2	27.2	28.1	28.8	29.4	29.9	30.1	30.4	
3.0	6.0	8.9	11.0	13.0	15.0	17.1	19.4	20.5	21.5	22.4	23.3	24.2	25.2	26.2	27.0	27.8	28.6	29.3	29.5	
3.1	6.2	9.2	12.2	14.2	16.2	18.2	20.6	22.8	24.6	26.6	27.6	28.6	29.6	30.3	30.9	31.5	31.8	32.0	32.0	
3.2	6.4	9.5	12.5	14.5	16.5	18.1	20.1	22.2	24.6	26.6	27.5	28.5	29.6	30.6	31.3	31.9	32.2	32.4	32.4	
3.3	6.6	9.8	13.3	15.3	17.3	19.4	21.4	23.4	25.4	27.4	29.4	31.4	33.4	35.4	37.4	39.4	39.7	39.8	39.9	
3.4	6.8	10.1	13.7	15.7	17.6	19.5	21.5	23.5	25.4	27.4	29.4	31.4	33.4	35.4	37.4	39.4	40.1	40.4	40.6	
3.5	7.0	10.4	14.1	16.1	18.1	19.8	21.8	23.8	25.7	27.7	29.7	31.7	33.7	35.7	37.7	39.7	39.8	39.8	39.9	
3.6	7.2	10.7	14.1	17.3	20.3	23.0	25.3	27.3	29.3	31.3	33.3	35.3	37.3	39.3	41.3	43.3	45.3	45.5	45.5	
3.7	7.4	11.0	14.0	17.2	20.8	23.5	26.2	28.0	29.7	31.7	33.7	35.7	37.7	39.7	41.7	43.7	45.7	45.7	45.7	
3.8	7.6	11.3	14.9	18.2	21.3	24.1	27.4	29.6	31.4	33.4	35.4	37.4	39.4	41.4	43.4	45.4	47.4	47.4	47.4	
3.9	7.8	11.6	15.3	18.7	21.9	25.3	27.8	29.9	31.1	33.2	35.2	37.2	39.2	41.2	43.2	45.2	47.2	47.2	47.2	
4.0	8.0	11.9	15.9	18.7	21.2	24.0	27.3	29.7	31.7	33.7	35.7	37.7	39.7	41.7	43.7	45.7	47.7	47.7	47.7	
4.1	8.2	12.2	16.0	19.6	22.9	25.4	28.4	30.9	32.1	34.1	36.1	38.1	40.1	42.1	44.1	46.1	48.1	48.1	48.1	
4.2	8.4	12.5	16.4	20.1	23.4	26.9	29.0	31.2	33.1	35.1	37.1	39.1	41.1	43.1	45.1	47.1	49.1	49.1	49.1	
4.3	8.6	12.8	16.8	20.5	24.0	27.6	30.1	32.6	34.6	36.6	38.6	40.6	42.6	44.6	46.6	48.6	48.6	48.6	48.6	
4.4	8.8	13.1	17.2	20.8	24.1	27.8	30.3	32.8	35.3	37.3	39.3	41.3	43.3	45.3	47.3	49.3	49.3	49.3	49.3	
4.5	9.0	13.3	17.5	21.5	25.0	28.1	30.8	33.1	35.5	37.5	39.5	41.5	43.5	45.5	47.5	49.5	49.5	49.5	49.5	
4.6	9.2	13.6	17.9	21.9	25.5	28.7	31.4	33.8	35.8	37.8	39.8	41.8	43.8	45.8	47.8	49.8	49.8	49.8	49.8	
4.7	9.4	13.9	18.3	22.4	26.0	29.3	32.0	34.4	36.3	38.3	40.3	42.3	44.3	46.3	48.3	50.3	50.3	50.3	50.3	
4.8	9.6	14.2	18.7	22.8	26.6	29.8	32.6	35.0	37.0	39.0	41.0	43.0	45.0	47.0	49.0	51.0	51.0	51.0	51.0	
4.9	9.8	14.5	19.1	23.3	27.1	30.4	33.2	35.6	37.6	39.6	41.6	43.6	45.6	47.6	49.6	51.6	51.6	51.6	51.6	
5.0	10.0	15.1	19.4	23.7	27.6	31.0	33.8	36.2	38.2	40.2	42.2	44.2	46.2	48.2	50.2	52.2	52.2	52.2	52.2	
5.1	10.2	15.4	19.8	24.2	28.1	31.5	34.4	36.8	38.8	40.4	41.7	42.7	43.7	44.7	45.7	46.7	47.7	47.7	47.7	
5.2	10.4	15.7	20.2	24.6	28.6	32.1	35.0	37.4	39.5	41.5	42.3	43.3	44.3	45.3	46.3	47.3	48.3	48.3	48.3	
5.3	10.6	16.0	20.9	25.5	29.6	33.2	36.2	38.1	40.1	41.7	43.0	44.0	45.0	46.0	47.0	48.0	49.0	49.0	49.0	
5.4	10.8	16.3	21.3	26.0	30.6	34.2	37.7	40.3	42.9	44.3	46.0	47.6	49.1	50.6	51.6	52.6	53.6	53.6	53.6	
5.5	11.0	16.6	21.7	26.4	30.6	34.4	37.3	40.9	43.5	45.1	46.7	48.2	49.7	51.2	52.7	54.2	55.2	55.2	55.2	
5.6	11.2	16.8	22.1	26.9	31.0	34.8	37.9	41.3	44.0	45.6	47.2	48.7	50.2	51.7	53.2	54.7	55.7	55.7	55.7	
5.7	11.3	16.9	22.4	27.3	31.1	34.8	37.9	40.5	43.5	45.1	46.7	48.2	49.7	51.2	52.7	54.2	55.2	55.2	55.2	
5.8	11.5	17.1	22.4	27.3	31.1	35.3	38.5	41.4	44.4	46.1	47.7	49.2	50.7	52.2	53.7	55.2	56.2	56.2	56.2	
5.9	11.7	17.4	22.8	27.8	32.1	35.9	39.0	41.6	44.7	46.7	48.5	50.1	51.7	53.2	54.7	56.2	57.2	57.2	57.2	

6.0	11.9	17.7	21.2	26.2	32.6	36.4	39.6	42.2	44.3	46.0	47.3	48.4	49.2	49.8	50.4	51.1	51.3	51.5	51.6
6.1	12.1	18.0	21.6	28.6	33.1	37.0	40.7	43.8	45.5	47.2	48.5	49.6	50.4	51.0	51.4	51.7	52.1	52.1	52.2
6.2	12.3	18.3	22.9	29.1	33.6	37.5	40.7	43.8	45.5	47.2	48.5	49.6	50.4	51.0	51.4	51.7	52.1	52.1	52.2
6.3	12.5	18.6	24.3	30.1	34.0	38.0	41.3	44.0	45.5	46.6	47.8	49.2	49.8	50.8	51.6	52.0	52.3	52.5	52.8
6.4	12.7	18.9	24.7	30.7	34.6	38.6	41.9	44.5	46.6	47.7	48.4	49.6	50.4	51.6	52.2	52.6	52.9	53.2	53.4
6.5	12.9	19.2	25.1	30.8	35.1	39.1	42.4	45.1	47.3	49.0	50.4	51.4	52.2	52.8	53.2	53.5	53.8	53.9	54.0
6.6	13.1	19.5	25.4	30.9	35.6	39.6	43.0	45.9	47.9	49.6	50.9	52.0	52.8	53.4	54.1	54.4	54.5	54.6	54.6
6.7	13.3	19.8	25.8	31.3	36.1	39.6	42.0	45.5	47.3	49.6	50.6	51.5	52.6	53.4	54.0	54.4	54.7	55.0	55.2
6.8	13.5	20.1	26.0	31.7	36.6	40.7	44.1	46.8	48.5	50.8	52.1	53.2	54.0	54.6	55.0	55.3	55.6	55.8	55.8
6.9	13.7	20.3	26.6	32.2	37.1	41.2	44.6	47.4	49.6	51.4	52.7	53.8	54.6	55.2	55.6	56.1	56.3	56.4	56.4
7.0	13.9	20.6	27.0	32.6	37.5	41.7	45.2	48.0	50.7	52.9	53.3	54.4	55.2	56.3	56.7	56.9	57.0	57.6	57.6
7.1	7.1	14.1	20.9	27.9	33.0	38.0	42.2	45.7	48.5	50.7	52.7	53.9	54.9	55.7	56.8	57.1	57.3	57.5	57.6
7.2	7.2	14.3	21.2	27.7	33.5	38.5	42.8	46.3	49.1	51.3	53.1	54.5	55.5	56.3	56.9	57.4	57.9	58.1	58.2
7.3	7.3	14.5	21.5	28.0	33.9	39.0	43.3	46.8	49.6	51.9	53.7	55.0	55.6	56.1	56.9	57.5	58.3	58.5	58.7
7.4	7.4	14.7	21.8	28.4	34.3	39.5	43.8	47.3	49.7	51.2	53.4	54.2	55.6	56.7	57.5	58.1	58.8	59.1	59.3
7.5	7.5	14.9	22.1	28.8	34.8	40.4	44.3	47.9	50.7	53.0	54.8	56.2	57.2	58.6	59.2	59.7	60.0	60.4	60.5
7.6	7.6	15.1	22.4	29.1	35.6	41.1	44.6	48.4	51.3	53.6	55.4	56.7	57.8	58.6	59.2	59.7	60.1	60.4	60.5
7.7	7.7	15.3	22.7	29.5	35.6	41.4	45.3	49.0	51.8	54.1	55.9	57.3	58.4	59.2	59.8	60.4	60.8	61.1	61.6
7.8	7.8	15.5	23.2	29.9	36.1	41.4	45.9	49.5	52.4	54.7	56.5	57.9	59.0	59.5	60.1	60.7	61.3	61.5	61.6
7.9	7.9	15.7	23.2	30.2	36.5	41.9	46.4	50.0	52.9	55.2	57.0	58.4	59.5	60.3	60.9	61.5	62.1	62.2	62.2
8.0	8.0	15.9	23.5	30.6	36.9	42.4	46.9	50.5	52.5	55.5	57.8	59.0	59.5	60.3	60.9	61.5	62.5	62.6	62.7
8.1	8.1	16.1	23.8	31.0	37.4	42.8	46.3	49.7	51.6	54.0	56.3	58.2	59.6	60.6	61.5	62.1	62.5	63.0	63.2
8.2	8.2	16.3	24.1	31.3	37.8	43.3	46.8	50.3	52.1	54.6	56.9	58.7	60.1	61.7	62.6	63.1	63.6	63.8	63.9
8.3	8.3	16.5	24.4	31.7	38.2	43.8	47.4	50.9	52.1	54.7	57.4	59.3	60.7	61.7	62.6	63.2	63.6	64.2	64.4
8.4	8.4	16.7	24.7	31.9	38.6	43.8	47.9	51.4	53.6	56.2	58.0	59.8	61.2	62.3	63.1	63.7	64.2	64.5	65.0
8.5	8.5	16.9	25.0	32.4	39.1	44.7	49.4	53.2	55.2	58.5	60.4	61.8	62.9	64.1	65.0	66.3	66.5	66.5	66.5
8.6	8.6	17.1	25.3	31.8	38.2	43.8	47.5	51.2	53.7	56.7	59.1	60.9	62.3	63.4	64.2	65.3	65.6	65.8	66.0
8.7	8.7	17.3	24.5	31.2	39.9	45.7	50.4	54.2	57.2	59.5	61.4	62.9	64.9	66.4	67.6	68.6	69.4	69.6	69.9
8.8	8.8	17.5	25.8	31.5	39.3	46.1	50.9	54.7	57.8	60.1	62.0	63.4	64.5	65.3	66.4	67.1	67.2	67.4	67.5
8.9	8.9	17.7	26.1	31.9	39.8	46.8	51.4	55.2	58.3	60.7	62.5	63.9	65.0	66.4	67.5	68.0	68.3	68.4	68.5
9.0	9.0	17.9	26.4	32.3	39.3	47.1	51.9	55.7	58.8	61.2	63.1	64.5	65.6	66.4	67.6	68.3	68.8	69.2	69.3
9.1	9.1	18.1	26.7	32.6	39.6	47.5	52.4	56.3	59.3	61.7	63.6	65.0	66.1	67.5	68.6	69.1	69.7	70.0	70.4
9.2	9.2	18.3	27.0	32.9	39.9	47.0	51.6	55.0	58.9	61.9	63.8	65.2	66.4	67.7	68.9	69.5	70.1	70.4	70.9
9.3	9.3	18.5	27.3	35.3	42.5	48.5	53.8	57.8	60.9	63.2	64.7	66.1	67.2	68.5	69.7	70.1	70.7	71.1	71.5
9.4	9.4	18.7	27.5	35.7	42.9	48.9	53.8	57.8	60.9	63.2	64.5	65.2	66.6	67.8	69.2	69.6	70.1	70.5	70.9
9.5	9.5	18.9	27.8	36.1	43.3	49.4	54.3	58.3	61.4	63.8	65.7	67.2	68.8	69.7	70.5	70.7	71.1	71.4	71.5
9.6	9.6	19.1	28.1	36.4	43.7	49.8	54.8	58.8	61.9	64.4	66.2	67.7	69.3	70.1	70.8	71.2	71.7	71.9	72.0
9.7	9.7	19.3	28.4	36.8	44.1	50.3	55.3	59.3	62.4	64.9	66.8	68.2	69.7	70.6	71.3	72.3	72.7	73.2	73.4
9.8	9.8	19.5	28.7	37.2	44.5	50.8	55.8	59.8	62.9	65.4	67.3	68.7	69.8	70.7	71.8	72.8	73.2	73.6	73.6
9.9	9.9	19.6	29.0	37.5	45.0	51.2	56.3	60.3	63.5	65.9	67.8	69.3	70.4	71.2	72.3	73.1	73.5	73.9	73.9
10.0	10.0	19.8	29.3	37.9	45.4	51.7	56.8	60.8	64.0	66.4	68.3	69.8	70.9	71.7	72.8	73.1	73.5	73.9	73.9
10.1	10.1	20.0	19.9	39.7	57	72	84	93	100	112	115	116	117	118	119	119	119	119	119
10.2	10.2	20.2	20.0	39.8	58	83	104	119	130	137	143	147	150	152	154	156	156	156	156
10.3	10.3	20.4	20.2	39.9	59	96	106	134	151	163	171	177	181	184	186	188	189	191	192
10.4	10.4	20.6	20.4	40.0	60	114	114	134	151	161	171	194	208	213	218	221	222	223	224
10.5	10.5	20.8	20.6	40.2	62	115	115	135	152	161	171	189	209	223	238	242	245	251	253
10.6	10.6	21.0	20.8	40.4	64	116	116	136	153	161	171	187	207	220	235	247	249	251	253
10.7	10.7	21.2	21.0	40.6	66	117	117	137	154	162	171	188	207	221	236	250	275	280	281
10.8	10.8	21.4	21.2	40.8	68	118	118	138	155	163	171	189	208	222	237	265	275	280	281
10.9	10.9	21.6	21.4	41.0	70	119	119	139	156	164	171	189	208	223	238	265	275	280	281
11.0	11.0	21.8	21.6	41.2	72	120	120	140	157	165	172	190	209	224	239	266	276	281	281
11.1	11.1	22.0	21.8	41.4	74	121	121	141	158	166	173	191	210	225	240	267	277	282	282
11.2	11.2	22.2	22.0	41.6	76	122	122	142	159	167	174	192	211	226	241	268	278	283	283
11.3	11.3	22.4	22.2	41.8	78	123	123	143	160	168	175	193	211	227	242	269	279	284	284
11.4	11.4	22.6	22.4	42.0	80	124	124	144	161	169	176	194	211	228	243	270	280	285	285
11.5	11.5	22.8	22.6	42.2	82	125	125	145	162	170	177	195	211	229	244	271	281	286	286
11.6	11.6	23.0	22.8	42.4	84	126	126	146	163	171	178	196	211	229	245	272	282	287	287
11.7	11.7	23.2	23.0	42.6	86	127	127	147	164	172	179	197	211	229	246	273	283	288	288
11.8	11.8	23.4	23.2	42.8	88	128	128	148	165	173	180	198	211	229	247	274	284	289	289
11.9	11.9	23.6	23.4	43.0	90	129	129	149	166	174	181	199	211	229	248	275	285	290	290
12.0	12.0	23.8	23.6	43.2	92	130	130	150	167	175	182	200	211	229	249	276	286	295	295
12.1	12.1	24.0	23.8	43.4	94	131	131	151	168	176	183	201	211	229	250	277	287	296	296
12.2	12.2	24.2	24.0	43.6	96	132	132	152	169	177	184	201	211	229	251	278	288	297	297
12.3	12.3	24.4	24.2	43.8	98	133	133	153	170	178	185	201	211	229	253	280	298	307	307
12.4	12.4	24.6	24.4	44.0	100	134	134	154	171	179	186	201	211	229	254	281	299	307	307
12.5	12.5	24.8	24.6	44.2	102	135	135	155	172	180	187	201	211	229	256	282	299	307	307
12.6	12.6	25.0	24.8	44.4	104	136	136	156	173	181	188	201	211	229	258	284	300	308	308
12.7	12.7	25.2	25.0	44.6	106	137	137	157	174	182	189	201	211	229	259				

FIG. 2. Plot of  $F(Sc)$  vs.  $Sc$  or  $F(Pr)$  vs.  $Pr$ .

numbers from 0.5 to 10 000. Fig. 2 shows the calculated values of the function  $F(Sc, 20)$  or  $F(Pr, 20)$ . For rapid calculation purposes the function shown on Fig. 2 can be approximated with a maximum error of  $\pm 2$  per cent as follows:

For turbulent diffusion in gases the function  $F(Sc, 20)$  can be represented by the equation

$$F(Sc, 20) = 13.0 (Sc)^{0.80}, \text{ where } 0.2 \leq Sc \leq 2. \quad (17)$$

For diffusion in liquid streams the following equation is proposed:

$$F(Sc, 20) = 17.25 (Sc)^{0.66}, \quad \text{where } 100 \leq Sc \leq 10000. \quad (18)$$

For the intermediate range of Schmidt and Prandtl numbers the function  $F$  can be given by the equation

$$F(Sc, 20) = 13.8 (Sc)^{0.71}, \text{ where } 2 \leq Sc \leq 100. \quad (19)$$

In Figs. 3 and 4 the proposed mass- and heat-transfer Stanton-number relationship is compared with heat- and mass-transfer data of many workers at Reynolds numbers of 10 000 and 25 000. The experimental points representing a mean through the data are the ones chosen by Deissler except for the datum point of Meyerink and Friedlander [11] which was taken from their mean curve through the data. The proposed expression agrees very well with the data over a wide range of Schmidt and Prandtl numbers.

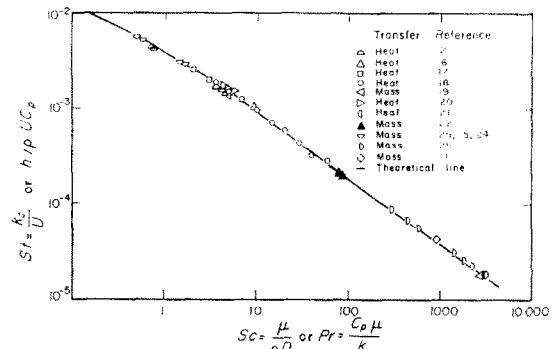


FIG. 3. Comparison of present theory with experimental data at a Reynolds number of 10 000.

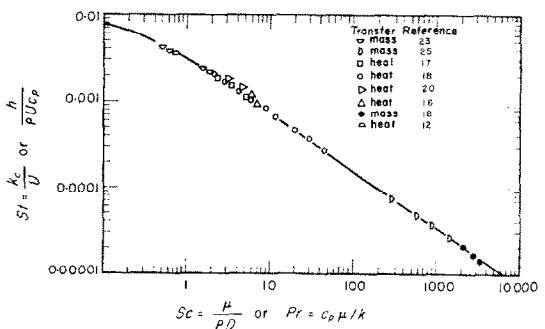


FIG. 4. Comparison of present theory with experimental data at a Reynolds number of 25 000.

Fig. 5 shows a comparison of the values calculated from the proposed Stanton number relationship at a Reynolds number of 50 000 with those calculated from the expressions of Deissler [10], Gowariker [6], Lin, Moulton and Putnam [13], Rannie [7], and von Kármán [14]. It is noted that von Kármán's relationship

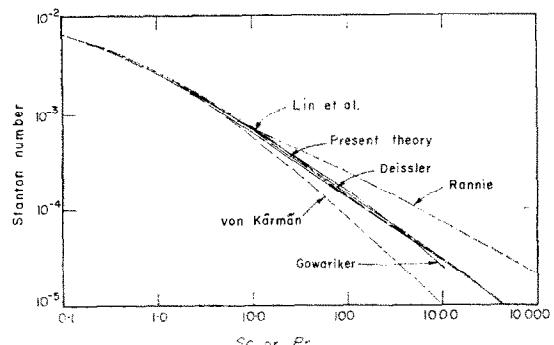


FIG. 5. Comparison of several analogies at a Reynolds number of 50 000.

gives too low results and Rannie's expression gives too high results at large Schmidt and Prandtl numbers. However, our correlation agrees very well with those of Lin, Moulton and Putnam, and Deissler.

By combinations of (12) and (18), we express Stanton number in terms of Schmidt number and friction factor for large values of Schmidt number by

$$St_m = 0.058 \sqrt{(f/2)} (Sc)^{-0.66} \quad (20)$$

It is of interest that the exponent on the Schmidt number as given by (20) is in agreement with the Chilton and Colburn analogy [15].

The curves for heat and mass transfer expressed in the form of Nusselt numbers are shown in Fig. 6. The Nusselt number for mass transfer (i.e. the Sherwood number) can be correlated by the following expressions:

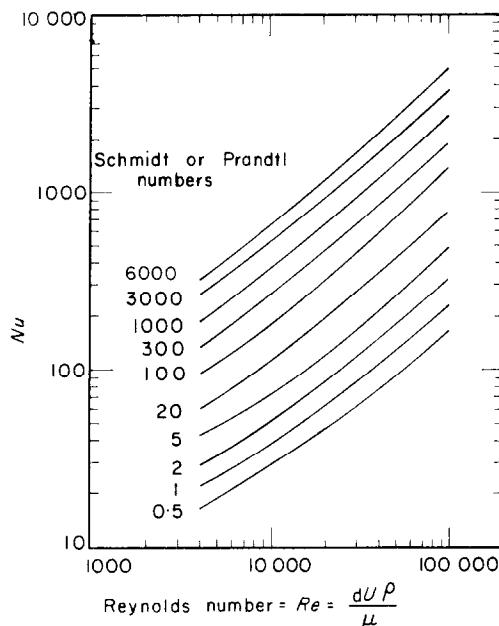


FIG. 6. Nusselt number vs. Reynolds number at various Schmidt or Prandtl numbers.

$$Nu = \frac{(f/2)(Re)(Sc)}{1 + \sqrt{(f/2)[13.0(Sc)^{0.80} - 13.0]}} \quad \text{for } 0.2 \leq Sc \leq 2, \quad (21)$$

$$Nu = \frac{(f/2)(Re)(Sc)}{1 + \sqrt{(f/2)[13.8(Sc)^{0.71} - 13.0]}} \quad \text{for } 2 \leq Sc \leq 100, \quad (22)$$

$$\text{and } Nu = (0.058) \sqrt{(f/2)} (Re) (Sc)^{0.34} \quad \text{for } 100 \leq Sc \leq 10,000. \quad (23)$$

Heat transfer Nusselt numbers can be obtained by replacing Schmidt numbers by Prandtl numbers in the above expressions. These expressions for Nusselt numbers are based on the difference between wall and average concentration or temperature.

## CONCLUSIONS

We show that agreement between predicted values and experimental data of mass- and heat-transfer rates supports the use of the proposed eddy-viscosity distribution function. Our predicted transfer rates agree very well with those calculated from Lin, Moulton and Putnam relationships, which are in excellent agreement with experimental data. Our proposed eddy-viscosity expression is useful because it applies to the whole wall region of pipe flow and is developed on a sound theoretical basis. It would also appear that the previous concept of sharply defined fluid layers is not necessary.

To calculate mass- and heat-transfer rates, simplified equations in the form of equations (17), (18) and (19) can be used to predict the complicated functions  $F(Sc)$  and  $F(Pr)$ . Also, simplified equations in the form of equations (21), (22) and (23) are proposed to predict heat- and mass-transfer Nusselt-number relationships.

## REFERENCES

1. A. FAGE and H. C. TOWNSEND, *Proc. Roy. Soc. London* **135**, 656 (1932).
2. T. J. HANRATTY, *A.I.Ch.E.J.* **2**, 359 (1956).
3. T. K. SHERWOOD, *Chem. Engng. Progr. Symposium Series*, **55**, 71 (1959).
4. D. T. WASAN, C. L. TIEN and C. R. WILKE, *A.I.Ch.E.J.* **9**, 567 (1963).
5. A. A. TOWNSEND, *The Structure of Turbulent Shear Flow*, Cambridge University Press (1956).
6. V. R. GOWARIKER, U.K. Atom. Energy Comm. AERE 1055 (1962).
7. W. D. RANNIE, *J. Aero. Sci.* **23**, 485 (1956).
8. C. A. SLEICHER, *Trans. ASME* **80**, 693 (1958).
9. J. LAUFER, Natl. Advisory Comm. Aeron. Tech. Report 1174 (1954).
10. R. G. DEISSLER, Natl. Advisory Comm. Aeron. Tech. Report 1210 (1955).
11. E. S. C. MEYERINK and S. K. FRIEDLANDER, National Science Foundation Report Grant No. G5079 (May 1960).

12. R. G. DEISSLER and C. S. EIAN, Natl. Advisory Comm. Aeron. TN 2629 (1952).
13. C. S. LIN, R. W. MOULTON and G. L. PUTNAM, *Industr. Engng. Chem.* **45**, 636 (1953).
14. T. VON KÁRMÁN, *Trans. ASME* **61**, 705 (1939).
15. T. H. CHILTON and A. P. COLBURN, *Industr. Engng. Chem.* **26**, 1183 (1934).
16. A. E. EAGLE and R. M. FERGUSON, *Proc. Roy. Soc. London* **127**, 540 (1930).
17. F. KREITH and M. SUMMERFIELD, *Trans. ASME* **72**, 869 (1950).
18. E. BERNARDO and S. EIAN, Natl. Advisory Comm. Aeron. WRE 136 (1945).
19. W. H. LINTON and T. K. SHERWOOD, *Chem. Engng. Progr.* **46**, 258 (1950).
20. M. D. GRELE and L. GREDEON, Natl. Advisory Comm. Aeron. RM E 53Lo9 (1953).
21. H. W. HOFFMAN, Oakridge Natl. Laboratory Report 1370, Contract No. W-7405-eng-26 (1952).
22. C. F. BONILLA, NYO-3086, U.S. Atom. Energy Comm. AEC Contract No. AT 30-1-1100 (1951).
23. W. I. BARNET and K. A. KOBE, *Industr. Engng. Chem.* **33**, 436 (1941).
24. M. L. JACKSON and N. H. CEGLSKE, *Industr. Engng. Chem.* **42**, 1188 (1950).
25. C. S. LIN, E. B. DENTON, H. S. GASKIL and G. L. PUTNAM, *Industr. Engng. Comm.* **43**, 2136 (1951).
26. S. J. KAUFMAN and F. D. SELY, Natl. Advisory Comm. Aeron. RME 50G 31 (1950).

**Résumé**—Les auteurs présentent une nouvelle expression du transport de chaleur et de masse pour un fluide en écoulement pleinement turbulent dans une conduite. Cette expression diffère des relations empiriques précédentes en ce qu'elle est établie à partir d'une distribution de viscosité turbulente continue de la paroi vers le centre de la conduite. Les coefficients de transport calculés à partir de la nouvelle corrélation sont en excellent accord avec les données expérimentales de transport de chaleur et de masse dans les écoulements fluides.

**Zusammenfassung**—Zum Beschreiben des Stoff- und Wärmeüberganges an ein Fluid in voll entwickelter turbulenter Rohrströmung wird eine neue Beziehung angegeben. Von älteren empirischen Beziehungen unterscheidet sie sich dadurch, dass sie auf einer theoretischen kontinuirlichen Verteilung der Wirbelviskosität von der Rohrwand bis zur Achse beruht. Mittels dieser neuen Beziehung berechnete Übergangswerte stimmen mit Versuchsergebnissen ausgezeichnet überein.

**Аннотация**—Дается новое соотношение для описания процесса переноса тепла и вещества к жидкости при полностью развитом турбулентном течении в трубе. Это соотношение отличается от предыдущих эмпирических соотношений тем, что оно основано на теоретическом распределении турбулентной вязкости от стени к центру трубы. Скорости переноса, вычисленные с помощью новой корреляции, хорошо согласуются с экспериментальными данными по переносу тепла и массы к потокам жидкости.